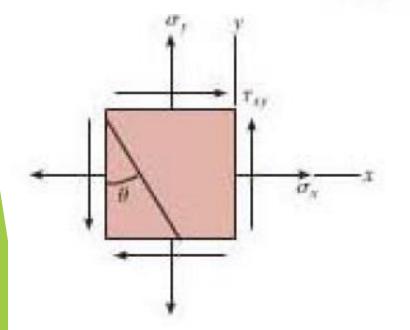
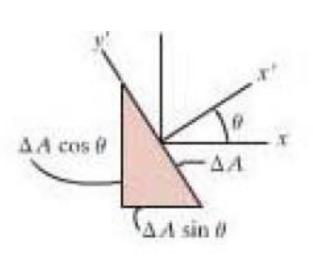
Strength of Material Equations of Plane-Stresses Transformation

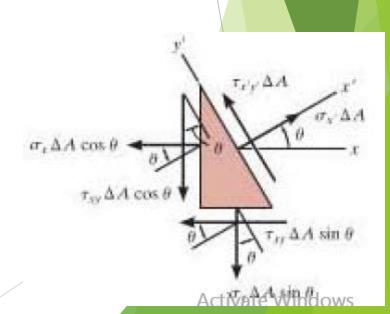
Plane-Stresses Transformation

The method of transforming of the normal and shear stress components from the x, y to the x" y" coordinate axes, as discussed in the previous section, can be developed in a general manner and expressed as a set of stress-transformation equations.

 $+ \angle \Sigma F_{x'} = 0; \quad \sigma_{x'} \Delta A - (\tau_{xy} \Delta A \sin \theta) \cos \theta - (\sigma_{y} \Delta A \sin \theta) \sin \theta$ $- (\tau_{xy} \Delta A \cos \theta) \sin \theta - (\sigma_{x} \Delta A \cos \theta) \cos \theta = 0$ $\sigma_{x'} = \sigma_{x} \cos^{2} \theta + \sigma_{y} \sin^{2} \theta + \tau_{xy} (2 \sin \theta \cos \theta)$ $+ \nabla \Sigma F_{y'} = 0; \quad \tau_{x'y'} \Delta A + (\tau_{xy} \Delta A \sin \theta) \sin \theta - (\sigma_{y} \Delta A \sin \theta) \cos \theta$ $- (\tau_{xy} \Delta A \cos \theta) \cos \theta + (\sigma_{x} \Delta A \cos \theta) \sin \theta = 0$ $\tau_{x'y'} = (\sigma_{y'} - \sigma_{x'}) \sin \theta \cos \theta + \tau_{xy} (\cos^{2} \theta - \sin^{2} \theta)$







These two equations may be simplified by using the trigonometric identities $\sin 2\theta = 2 \sin \theta \cos \theta$, $\sin^2 \theta = (1 - \cos 2\theta)/2$, and $\cos^2 \theta = (1 + \cos 2\theta)/2$, in which case,

$$\sigma_{x'} = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta$$

$$\tau_{x'y'} = \frac{\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta$$

If the normal stress acting in the y' direction is needed, it can be obtained by simply substituting $(\theta = \theta + 90^{\circ})$ for θ into Eq. 9-1,

$$\sigma_{y'} = \frac{\sigma_x + \sigma_y}{2} - \frac{\sigma_x - \sigma_y}{2} \cos 2\theta - \tau_{xy} \sin 2\theta$$

The state of plane stress at a point is represented by the element shown in Fig. 9-7a. Determine the state of stress at the point on another element oriented 30° clockwise from the position shown.

$$\sigma_x = -80 \text{ MPa}$$

$$\sigma_v = 50 \text{ MPa}$$

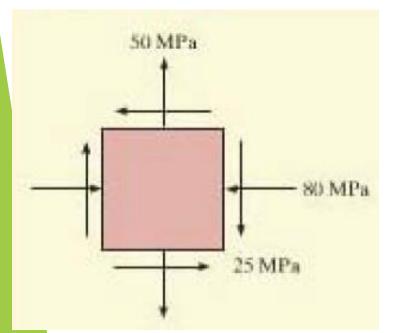
$$\sigma_x = -80 \text{ MPa}$$
 $\sigma_y = 50 \text{ MPa}$ $\tau_{xy} = -25 \text{ MPa}$

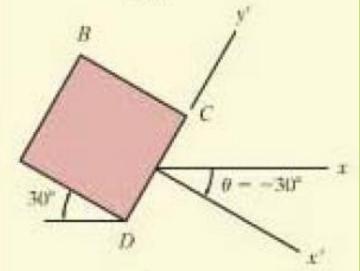
The plane DC

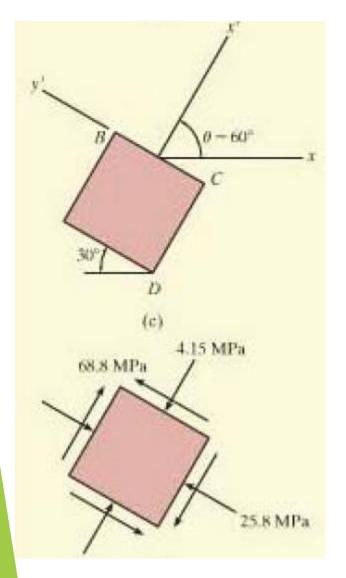
$$\sigma_{x'} = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta$$

$$= \frac{-80 + 50}{2} + \frac{-80 - 50}{2} \cos 2(-30^\circ) + (-25) \sin 2(-30^\circ)$$

$$= -25.8 \text{ MPa}$$
Activate Windows







$$\tau_{x'y'} = -\frac{\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta$$

$$= -\frac{-80 - 50}{2} \sin 2(-30^\circ) + (-25) \cos 2(-30^\circ)$$

$$= -68.8 \text{ MPa}$$
Ans.

The plane BC

$$\sigma_{x'} = \frac{-80 + 50}{2} + \frac{-80 - 50}{2} \cos 2(60^{\circ}) + (-25) \sin 2(60^{\circ})$$

$$= -4.15 \text{ MPa}$$

$$\tau_{x'y'} = -\frac{-80 - 50}{2} \sin 2(60^{\circ}) + (-25) \cos 2(60^{\circ})$$

$$= 68.8 \text{ MPa}$$
Ans.